



# STABILIZATION OF THE PROGRAMMED MOTIONS OF A RIGID BODY WITH UNCERTAINTY IN THE PARAMETERS OF THE EQUATIONS OF DYNAMICS†

A. T. ZAREMBA

St Petersburg

(Received 26 August 1996)

A control scheme is proposed to guarantee the asymptotic stability of a given programmed motion of a rigid body rotating about a fixed point. The body is controlled by means of a couple of reactive forces, or the control action is created by rotating flywheels. The inertial parameters and angular momentum of the system are estimated while the motion is in progress. The control is synthesized by expressing the equations of dynamics in a form that is linear in the parameter vector and by using the passivity property of the dynamical object. A law is proposed for the control and for adjusting the parameters that guarantees the asymptotic stability of the motion and, for programmed motions that satisfy the condition of non-vanishing action, guarantees the convergence of the vector of adjusted parameters to its true value. The domain in phase space for which exponential stabilization is achieved is determined. © 1997 Elsevier Science Ltd. All rights reserved.

An analogous problem has been considered previously [1]; the control law obtained there for the rotary motion of an unmanned aircraft requires the computation of complicated non-linear expressions, including the Jacobian of the kinematic relations and its derivatives, which are not well-conditioned for rotations close to  $\pm\pi$ .

Unlike those results, the algorithms proposed below have no singularities anywhere in phase space.

A dynamical control of a rigid body for certain inertial parameters was constructed in [2]. Recurrent algorithms for setting up the equations of dynamics have been used to synthesize a control that ensures stabilization of the programmed motion of a multiple chain of rigid bodies [3].

## 1. DYNAMICAL MODEL OF THE SYSTEM

We will consider a rigid body  $P_0$  with a fixed point  $O$  at its centre of mass. Together with  $P_0$  we consider a system of coordinates  $Oxyz$  attached to the body and a triple of mutually orthogonal unit vectors  $c_1, c_2, c_3$  occupying a fixed position in  $Oxyz$ . A programmed motion is defined by a triple of mutually orthogonal unit vectors  $s_1, s_2, s_3$  rotating as a rigid body at angular velocity  $\omega_d$  relative to the absolute system of coordinates  $O\xi\eta\zeta$ .

Rotary motion of the body is described by Euler's equations of dynamics

$$M\dot{\omega} + \omega \times M\omega = u \quad (1.1)$$

where  $\omega$  is the angular velocity of the controlled body and  $u \in R^3$  is a vector of control torques created by couples of reactive forces.

If  $P_0$  is controlled by means of rotating symmetric flywheels  $P_1, \dots, P_m$ , then, when there are no external forces, the equations of motion of the system of bodies may be written as follows [2, 4]:

$$M\dot{\omega} + \omega \times R^t L_0 = u, \quad L_0 = \text{const}, \quad M = \theta_0 - \sum_{i=1}^m C_i b_i b_i^t, \quad u = \sum_{i=1}^m b_i \tau_i \quad (1.2)$$

where  $\theta_0$  is the inertia tensor of the system of bodies,  $b_i$  is a unit vector along the axis of rotation of the  $i$ th flywheel (the superscript  $t$  denotes transposition),  $C_i$  is the moment of inertia of the  $i$ th flywheel about its axis of rotation,  $L_0$  is the angular momentum of the system of bodies in the system of coordinates

†*Prikl. Mat. Mekh.* Vol. 61, No. 1, pp. 37–43, 1997.

$O\xi\eta\zeta, R \in SO(3)$  is the matrix of a rotation transforming the axes  $O\xi\eta\zeta$  into axes  $Oxyz$ , and  $\tau_i$  is a control torque applied to the axis of the  $i$ th flywheel.

Henceforth we will use a common notation for (1.1) and (1.2)

$$M\dot{\omega} + \omega \times N(\omega, L_0) = u \quad (1.3)$$

and consider the equation along with the Poisson kinematic equations

$$\dot{s}_i = -(\omega - \omega_d) \times s_i \quad (i = 1, 2, 3) \quad (1.4)$$

The inertial parameters and angular momentum of the dynamical object (1.3), (1.4) are unknown. It is required to determine a control torque  $u$  and adjustment law for the control parameters of the system so that the axes  $c_1, c_2, c_3$  attached to the body will stabilize in the direction of the corresponding fixed axes  $s_1, s_2, s_3$  and the angular velocity  $\omega$  will tend to  $\omega_d$ .

To determine the control law we will use the following property of the dynamical object (1.3).

1. The equation of dynamics (1.3) can be expressed in a form that is linear in the vector of unknown parameters  $\theta \in R^p$

$$M\dot{\omega}_r + \omega_r \times N(\omega, L_0) = Y(\omega, \omega_r, \dot{\omega}_r)\theta \quad (1.5)$$

where  $\omega_r, \dot{\omega}_r \in R^3$  are arbitrary vectors, the vector  $\theta$  contains the unknown components of the inertia matrix and the angular momentum, and  $Y$  is a known  $3 \times p$  matrix-valued function.

If the control is established by means of a couple of reactive forces and  $Oxyz$  are the principal central axes of inertia of  $P_0$ , then  $N(\omega, L_0) = M\omega$ ,  $\theta = (A, B, C)'$  and the  $3 \times 3$  matrix  $Y$  is

$$Y = \begin{vmatrix} \dot{\omega}_{r1} & -\omega_2\omega_{r3} & \omega_{r2}\omega_3 \\ \omega_1\omega_{r3} & \dot{\omega}_{r2} & -\omega_{r1}\omega_3 \\ -\omega_1\omega_{r2} & \omega_{r1}\omega_2 & \dot{\omega}_{r3} \end{vmatrix}$$

The following passivity property of the dynamical object is useful in finding the Lyapunov function used in synthesizing the control.

2. The differential equation (1.3) defines a passive mapping  $u \rightarrow \omega$ , that is

$$\int_0^T u' \omega d\tau \geq -\gamma^2 \text{ for all } T > 0 \text{ and some } \gamma \in R. \quad (1.6)$$

To prove inequality (1.6), we find

$$\int_0^T \omega' (M\dot{\omega} + \omega \times N(\omega, L_0)) d\tau = \frac{1}{2} (\omega'(T)M\omega(T) - \omega'(0)M\omega(0)) \geq -\gamma^2$$

We now introduce the normed spaces of square-integrable functions

$$L_2^n(R_+) = \left\{ x: R_+ \rightarrow R^n \left| \left( \int_0^\infty |x(t)|^2 dt \right)^{1/2} < \infty \right. \right\}$$

and bounded functions

$$L_\infty^n(R_+) = \left\{ x: R_+ \rightarrow R^n \left| \sup_{t \in [0, \infty)} |x(t)| < \infty \right. \right\}$$

where  $R_+$  is the set of non-negative real numbers.

**Lemma 1.** Let  $x: R_+ \rightarrow R^n$  be a real piecewise-smooth vector-valued function such that  $x \in L_2^n \cap L_\infty^n$  and  $\dot{x} \in L_\infty^n$ . Then  $x \rightarrow 0$  as  $t \rightarrow \infty$ .

*Proof.* It follows from the assumptions of the lemma that the function  $\varphi = x': R_+ \rightarrow R^1$  satisfies the conditions

$$\dot{\phi} = 2x'x \in L_\infty \text{ and } \int_0^\infty \phi dt = \int_0^\infty x'x dt \leq \text{const} \quad (1.7)$$

Consequently,  $\phi$  is in  $L_1$  and is also uniformly continuous on  $R_+$ . Therefore [5]  $\phi \rightarrow 0$  as  $t \rightarrow \infty$  and so also  $x \rightarrow 0$  as  $t \rightarrow \infty$ .

## 2. SYNTHESIS OF THE DYNAMIC CONTROL WITH PARAMETER ADJUSTMENT

We now define vector-valued functions characterizing the tracking errors of the programmed motion

$$e = \sum k_i s_i \times c_i, \quad \omega_r = \omega_d - \Lambda e, \quad v = \omega - \omega_r \quad (2.1)$$

The vector  $e \in R^3$  characterizes the orientation error of the rigid body;  $k_i$  are pairwise distinct positive numbers ( $i = 1, 2, 3$ ). The vector  $v \in R^3$  takes the orientation error and angular velocity error into account:  $v = \omega_e + \Lambda e$ , where  $\omega_e = \omega - \omega_d$  and  $\Lambda > 0$ . In all cases summation is from  $i = 1$  to  $i = 3$ .

Let  $\theta_r \in R^p$  be a parameter-estimate vector for the object and  $\theta_e \in R^p$ ,  $\theta_e = \theta_r - \theta$  the vector of parameter-estimate errors. It is assumed that the programmed motion satisfies the desired smoothness conditions  $\omega_d \in C^1(R_+)$ .

The problem is to define a control  $u(t)$  and a parameter-adjustment law so that the following control objective is attained

$$\lim_{t \rightarrow \infty} \omega = \omega_d, \quad \lim_{t \rightarrow \infty} s_i = c_i, \quad i = 1, 2, 3 \quad (2.2)$$

We will also define an auxiliary control objective

$$\lim_{t \rightarrow \infty} \omega = \omega_d, \quad \lim_{t \rightarrow \infty} e = 0 \quad (2.3)$$

which determines the convergence of the motions of the body to motions  $\omega = \omega_d$ ,  $s_1 = \pm c_1$ ,  $s_2 = \pm c_2$ ,  $s_3 = s_1 \times s_2$ , one of which corresponds to the programmed trajectory.

*Lemma 2.* For all  $T > 0$

$$\int_0^T \omega_e' e d\tau \geq -\gamma^2, \quad \gamma^2 = 2 \sum k_i \quad (2.4)$$

*Proof.* Substitution of (2.1) into (2.4) gives

$$\int_0^T \omega_e' e d\tau = \int_0^T \sum_{i=1}^3 k_i s_i \times c_i d\tau = \int_0^T \sum_{i=1}^3 k_i c_i' (\omega_e \times s_i) d\tau = - \int_0^T \sum_{i=1}^3 k_i c_i' \dot{s}_i d\tau = - \sum_{i=1}^3 k_i c_i' s_i \Big|_0^T \geq -\gamma^2$$

*Lemma 3.* If  $v \in L_2^3$ , then  $e, \omega_e \in L_2^3$ .

*Proof.* Using (2.1) and Lemma 2, we find that

$$\int_0^T \omega_e^2 d\tau + \int_0^T e^2 d\tau = \int_0^T v^2 d\tau - 2\Lambda \int_0^T \omega_e' e d\tau \leq \int_0^T v^2 d\tau + 2\Lambda \gamma^2 < \infty \quad (2.5)$$

Now let  $T \rightarrow \infty$ .

Consider the following Lyapunov function

$$V = [v' M v + \theta_e' \Gamma^{-1} \theta_e] / 2 \quad (2.6)$$

where  $\Gamma$  is the symmetric positive-definite  $3 \times 3$  matrix of feedback coefficients.

Differentiation of Eq. (2.6) along the trajectories of the system yields the control and parameter-adjustment law

$$u = M_r \dot{\omega}_r + \omega_r \times N(\omega, L_{0r}) - K_d v \quad (2.7)$$

$$\dot{\theta}_r = -\Gamma Y'(\omega, \omega_r, \dot{\omega}_r) v \quad (2.8)$$

where  $K_d$  is the symmetric positive-definite  $3 \times 3$  matrix of feedback coefficients and  $M_r$  and  $L_{0r}$  are estimates for the inertia tensor and angular momentum vector.

*Theorem 1.* Consider the motion of the rigid body (1.3), (1.4) subject to a control (2.7) and parameter-adjustment law (2.8). Then the bounded programmed trajectories  $\omega_d, \dot{\omega}_d \in L_\infty^3(R_+)$  achieve the auxiliary control objective (2.3) and the motions of the rigid body tend to asymptotically stable motions  $\omega = \omega_d, s_1 = \pm c_1, s_2 = \pm c_2, s_3 = s_1 \times s_2$ . The adjusted-parameter vector  $\theta_r$  and control  $u$  remain bounded.

*Proof.* Differentiation of Eq. (2.6), taking (1.3) and (1.4) into account, yields

$$\dot{V} = v' M(\dot{\omega} - \dot{\omega}_r) + \theta_e' \Gamma^{-1} \dot{\theta}_e = v'(u - \omega_r \times N(\omega, L_0) - M\dot{\omega}_r) + \theta_e' \Gamma^{-1} \dot{\theta}_e$$

Using the linear representation (1.5) and taking (2.7) and (2.8) into account, we obtain

$$\dot{V} = -v' K_d v \quad (2.9)$$

Here we have used the equality  $\dot{\theta}_e = \dot{\theta}_r$ , since  $\dot{\theta} = 0$ . It follows from (2.6) and (2.9) that

$$v \in L_2^3 \cap L_\infty^3, \quad \theta_e \in L_\infty^p$$

Using Lemma 3, we obtain  $\omega_e, e \in L_2^3$ , and it follows from the fact that  $e$  and  $v$  are bounded that  $\omega_e \in L_\infty^3$ .

Differentiation of Eq. (2.1) gives

$$\dot{e} = -\sum k_i (\omega_e \times s_i) \times c_i$$

that is, the boundedness of  $\omega_e$  implies that of  $\dot{e}$ . Consequently, by Lemma 1,  $e \rightarrow 0$  as  $t \rightarrow \infty$ .

We now prove that the vector  $v$  tends to zero. Since  $\omega_d \in L_\infty^3$ , the fact that  $\omega_e$  is bounded implies that  $\omega$  is bounded. Taking into account that  $\theta_e, \omega, \omega_d, \dot{e}$ , are bounded, we deduce from (2.7) that the control  $u$  is bounded. It then follows from (1.3) that  $\dot{\omega} \in L_\infty^3$ . Since  $\dot{\omega}$  is bounded, the same is true of  $\dot{v}$  and so, by Lemma 1,  $v \rightarrow 0$  as  $t \rightarrow \infty$ .

Since  $v$  and  $e$  tend to zero as  $t \rightarrow \infty$ , it follows that the motions of the rigid body tend to asymptotically stable motions  $\omega = \omega_d, s_1 = \pm c_1, s_2 = \pm c_2, s_3 = s_1 \times s_2$ , one of which corresponds to the programmed trajectory.

As an example, let us consider the problem of stabilizing a given position of a rigid body. Setting  $\omega_d = \dot{\omega}_d = 0$  in (2.7), we obtain

$$u = -K_d \omega - K_d \sum k_i s_i \times c_i - \Lambda Y(\omega, e, \dot{e}) \theta_r \quad (2.10)$$

This control differs from the stabilizing control of [2, p. 425] by dynamical terms proportional to the estimated parameters of the object, which improve the convergence of the orientation error to zero.

### 3. EXPONENTIAL STABILIZATION OF THE MOTION

To obtain exponential stabilization of a programmed motion and convergence of the estimated parameters of the object to their real values, we replace the Lyapunov function (2.6) by

$$V = 1/2 [v' M v + \theta_e' \Gamma^{-1} \theta_e + K_p \sum k_i (c_i - s_i)^2 + \|M v - \varepsilon - \varphi' \theta_e\|^2] \quad (3.1)$$

where  $\varepsilon \in R^3$  is an auxiliary vector and  $\varphi$  is a  $p \times 3$  matrix-valued function.

An additional term analogous to the last term in (3.1) was used in [6] for the exponential stabilization of the motion of a manipulator with rigid arms.

Differentiation of Eq. (3.1) gives the following control and parameter-adjustment law

$$u = Y_c \theta_r - K_p e - \alpha v, \quad \dot{\theta}_r = -\Gamma Y_c' v + \lambda_c \Gamma \varphi e \quad (3.2)$$

$$\dot{e} = -\lambda_c e - \alpha v - K_p e + \varphi' \Gamma Y_c' v - \lambda_c \varphi' \Gamma \varphi e, \quad \dot{\phi} = -\lambda_c \phi + Y_c'$$

where  $\lambda_c > 0$  and  $\alpha > \lambda_c (\sigma_M)^2$ ,  $\sigma_M$  being the largest eigenvalue of  $M$ ; the  $3 \times p$  matrix-valued function  $Y_c$  is defined by the equality

$$Y_c(\omega, \omega_r, \dot{\omega}_c) \theta = M \dot{\omega}_c + \omega_r \times N(\omega, L_0), \quad \dot{\omega}_c = \dot{\omega}_r - \lambda_c v \quad (3.3)$$

The matrix  $\Gamma$  in (3.2) is adjusted by using the method of least squares [7]

$$\dot{\Gamma}^{-1} = \frac{\lambda_c}{2} (\varphi \varphi' - \lambda(t) \Gamma^{-1}), \quad \Gamma(0) = \Gamma'(0) > 0, \quad \|\Gamma(0)\| \leq k_0; \quad \lambda(t) = \lambda_0 \left( 1 - \frac{\|\Gamma\|}{k_0} \right) \quad (3.4)$$

with a time-dependent parameter  $\lambda(t)$ .

The adjustment law (3.4) has the following properties: (a) for any  $t \geq 0$  we have inequalities  $\lambda(t) \geq 0$  and  $\Gamma(t) \leq k_0 I$ ; (b) for programmed motions with non-vanishing action, i.e. for which positive numbers  $\beta_1$ ,  $\beta_2$  and  $\delta$  such that

$$\beta_1 I_3 \leq \int_t^{t+\delta} \varphi'(\tau) \varphi(\tau) d\tau \leq \beta_2 I_3 \quad (3.5)$$

for any  $t \geq 0$ , a positive number  $\lambda_1 \geq 0$  exists such that  $\lambda(t) \geq \lambda_1$  for any  $t \geq 0$ .

If the variable  $\lambda(t)$  can be separated from zero, one can ensure that the parameter-estimate vector  $\theta_r$  will converge to its true value.

The control (3.2)–(3.4) is composite [7], since the parameter-adjustment operation uses both information about the tracking error of the programmed motion and the predicted value of the error.

**Theorem 2.** Consider the motion of a rigid body (1.3), (1.4) with the control and parameter-adjustment law (3.2)–(3.4). Then, if the programmed motions  $\omega_d, \dot{\omega}_d \in L_\infty^3(R_+)$  are bounded, the auxiliary control objective (2.3) will be achieved.

Moreover, if the programmed motion is such that the non-vanishing action condition (3.5) holds, then

- (a) the vector  $\theta$ , tends to the true value of the object parameters;
- (b) the vector  $e$  tends to zero,  $e \rightarrow 0$ ;
- (c) all steady motions  $\omega = \omega_d, s_1 = \pm c_1, s_2 = \pm c_2, s_3 = s_1 \times s_2, \theta_e = e = 0$  except the programmed motion are unstable, and the Lyapunov function (3.1) tends exponentially to zero in the domain

$$D: \{s: e'e > \Lambda^{-1} V_\delta\}, \quad \left( V_\delta = \sum_{i=1}^3 k_i (c_i - s_i)^2 \right) \quad (3.6)$$

*Proof.* Differentiation of the first term in (3.1), taking (1.3), (1.5) and the first equation of (3.2) into consideration, yields

$$\begin{aligned} T_1 &= v'(u - \omega \times N(\omega, L_0) - M \dot{\omega}_r) = v'(u - \omega_r \times N(\omega, L_0) - M \dot{\omega}_r) = \\ &= v'(u - Y_c \theta - \lambda_c M v) = -\lambda_c v' M v + v'(u - Y_c \theta_r + Y_c \theta_e) = \\ &= -\lambda_c v' M v - \alpha v' v - K_p v' e + v' Y_c \theta_e \end{aligned} \quad (3.7)$$

Differentiation of the second and third terms of (3.1) gives

$$T_2 + T_3 = \theta_e' \Gamma^{-1} \dot{\theta}_e + \frac{\lambda_c}{4} \left( \|\theta_e\|_{\varphi \varphi'}^2 - \lambda(t) \|\theta_e\|_{\Gamma^{-1}} \right) + K_p \omega_e' e \quad (3.8)$$

where  $\|X\|_A$  is the quadratic norm generated by a symmetric matrix  $A$ .

Differentiation of the last term in (3.1), taking (3.2) into consideration, gives [6]

$$\begin{aligned}
T_4 &= (Mv - \varepsilon - \varphi' \theta_e)' (-\lambda_c Mv + Y_c \theta_e - K_p e - \alpha v - \dot{\varepsilon} - \dot{\varphi}' \theta_e - \dot{\varphi}' \theta_e) = \\
&= \frac{\lambda_c}{2} \|Mv - \varepsilon - \varphi' \theta_e\|^2 - \frac{\lambda_c}{2} \|Mv - \varepsilon\|^2 + \lambda_c (Mv - \varepsilon)' \varphi' \theta_e - \frac{\lambda_c}{2} \|\theta_e\|_{\varphi\varphi'}^2 \leq \\
&\leq -\frac{\lambda_c}{2} \|Mv - \varepsilon - \varphi' \theta_e\|^2 - \frac{\lambda_c}{2} \|\theta_e\|_{\varphi\varphi'}^2 - \lambda_c \varepsilon' \varphi' \theta_e + \frac{\lambda_c}{2} \sigma_M \left( \mu \|v\|^2 + \frac{1}{\mu} \|\theta_e\|_{\varphi\varphi'}^2 \right) \quad (3.9)
\end{aligned}$$

Setting  $\mu = 2\sigma_M$  in (3.9) and adding equations (3.7)–(3.9) together, we obtain

$$\dot{V} \leq -\lambda_c \nu' Mv - \frac{\lambda_c}{2} \|Mv - \varepsilon - \varphi' \theta_e\|^2 - \frac{\lambda_c}{4} \lambda(t) \|\theta_e\|_{\Gamma^{-1}}^2 - K_p \Lambda e' e \quad (3.10)$$

The first part of the theorem now follows from (3.1) and (3.10).

We will now show that the parameter-estimate vector tends to its true value:  $\theta_e \rightarrow 0$ , provided condition (3.5) holds. Taking (3.5) into account, we deduce from (3.1) and (3.10) that  $\theta_e \in L^2_{\Gamma} \cap L^{\infty}_{\Gamma}$ .

We will now prove that  $\theta_e$  is bounded because of the second equality of (3.2). Hence  $\theta_e \rightarrow 0$  as  $t \rightarrow \infty$ .

Now  $\varepsilon \rightarrow 0$ , since  $\eta = Mv - \varepsilon - \varphi' \theta_e \rightarrow 0$ , which in turn follows from  $\eta \in L^3_2 \cap L^3_{\infty}$  (3.10) and the boundedness of the derivative  $\dot{\eta} \in L^3_{\infty}$  (3.2).

Using the convergence of the parameter-estimate vector to its true value, we will now show that all steady motions other than the programmed motion are unstable in Lyapunov's sense.

We will show, for example, that the steady motion  $\omega = \omega_d, s_1 = -c_1, s_2 = c_2, s_3 = -c_3, \theta_e = \varepsilon = 0$  is unstable. To that end, take the function

$$V_c = [v' Mv + \theta_e' \Gamma^{-1} \theta_e - k_1 (c_1 + s_1)^2 + k_2 (c_2 - s_2)^2 - k_3 (c_3 + s_3)^2] + \|Mv - \varepsilon - \varphi' \theta_e\|^2 / 2 \quad (3.11)$$

which is bounded in the domain  $V_c < 0$ , exists in an arbitrarily small neighbourhood of the steady motion  $\omega = \omega_d, s_1 = -c_1, s_2 = c_2, s_3 = -c_3, \theta_e = \varepsilon = 0$  for all  $t \geq t_0$  and has a negative-definite derivative (3.10). Now apply Chetayev's instability theorem [8].

The proof that the other steady motions are unstable is analogous. In the domain (3.6), inequality (3.10) may be written as follows:

$$\dot{V} + \gamma V \leq 0, \quad \gamma = \min\{\lambda_c, \lambda_c \lambda_1 / 2, 2\} > 0 \quad (3.12)$$

This relationship implies that the Lyapunov function tends exponentially to zero. This completes the proof of the theorem.

As an example, let us consider the problem of stabilizing the orientation of a rigid body in a given direction  $s_0(t)$ . Put  $e = s_0 \times c_0$  and  $V_{\delta} = (s_0 - c_0)^2$ , where  $c_0$  is a unit vector attached to the body. The first relationship for the controlling torque transforms to

$$\begin{aligned}
u &= M_r \dot{\omega}_d + \omega_d \times N(\omega, L_{0r}) - K_e e - K_v \omega_e - \Lambda Y(\omega, e, \dot{e}) \theta_r, \\
K_e &= (K_p + \alpha \Lambda) I_3 + \lambda_c \Lambda M_r, \quad K_v = \alpha I_3 + \lambda_c M_r
\end{aligned} \quad (3.13)$$

The domain (3.6) is

$$D: \{s_0: (s_0 c_0) > 2 / \Lambda - 1\} \quad (3.14)$$

On a sphere of unit radius about the point 0, this expression defines a domain with the following property: if the end of the unit vector  $s_0$  is within the domain, one obtains exponential stabilization of the motion provided condition (3.5) holds. If  $\Lambda > 1$ ,  $D_0$  encloses the equilibrium position  $s_0 = c_0$ , and as  $\Lambda$  decreases it tends to the unit sphere.

## REFERENCES

1. Slotine, J.-J. and Di Benedetto, M. D., Hamiltonian adaptive control of spacecraft. *IEEE Trans. Aut. Control*, 1990, 35(7), 848–852.
2. Zubov, V. I., *Lectures on Control Theory*. Nauka, Moscow, 1975.
3. Zaremba, A. T., Equation of dynamics of a multi-arm manipulator with holonomic constraints. *Izv. Akad. Nauk SSSR. MTT*, 1990, 4, 25–34.
4. Crouch, P. E., Spacecraft attitude control and stabilization: Application of geometric control theory to rigid body models. *IEEE Trans. Aut. Control*, 1984, 29(4), 321–331.
5. Desoer, C. M. and Vidyasagar, M., *Feedback Systems: Input–Output Properties*. Academic Press, New York, 1975.
6. Stotsky, A. A., Composite and pseudogradient algorithms for rigid robots. In *Proc. IEEE Workshop on "Variable Structure and Lyapunov Control of Uncertain Dynamical Systems"*, Sheffield, 1992.
7. Slotine, J.-J. and Li, W., Composite adaptive control of robot manipulators. *Automatica*, 1989, 25(4), 509–519.
8. Chetayev, N. G., *Stability of Motion*. Nauka, Moscow, 1990.

Translated by D.L.